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# Towards operational use of aircraft-derived observations: a case study at London Heathrow airport.

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#### 1 Abstract

Mode-Selective Enhanced Surveillance (Mode-S EHS) aircraft reports can be collected at a 2 low-cost, and are readily available around busy airports. The new work presented here demon-3 strates that observations derived from Mode-S EHS reports can be used to study the evolution 4 of temperature inversions since the data have a high spatial and temporal frequency. This 5 is illustrated by a case study centred around London Heathrow airport for the period 4 to 5 6 January 2015. Using Mode-S EHS reports from multiple aircraft and after applying quality 7 control criteria, vertical temperature profiles are constructed by aggregating these reports at 8 discrete intervals between the surface and 3000 m. To improve these derived temperatures, 9 four smoothing methods using low-pass filters are evaluated. The effect of smoothing reduces 10 the variance in the aircraft derived temperature by approximately half. After smoothing, the 11 temperature variance between the altitudes 3000 m and 1000 m is 1 K to 2 K; and below 1000 m 12 it is 2 K to 4 K. While the differences between the four smoothing methods are small, expo-13 nential smoothing is favoured because it uses all available Mode-S EHS reports. The resulting 14 vertical profiles may be useful in operational meteorology for identifying elevated temperature 15 inversions above 1000 m. However, below 1000 m they are less useful because of the reduced 16 precision of the reported Mach number. A better source of in situ temperature observations 17

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18 would be for aircraft to use the meteorological reporting function of their automatic dependent

19 surveillance (ADS) system.

# 20 **1** Introduction

Weather impacts on airports are an important problem for society (Ball et al., 2007; Markovic 21 et al., 2008; Barnhart et al., 2012). In particular, fog and low visibility conditions reduce the 22 air-traffic flow rates at airports as aircraft separations need to be increased to maintain safe 23 operations. The reduced flow rate increases costs in terms of the extra fuel that must be used, 24 loss of revenue due to reduced capacity at airports, environmental impacts on local air qual-25 ity and noise emissions, and climate impacts due to increased emissions of nitrogen oxides 26 and carbon dioxide (Mahashabde et al., 2011). Numerical weather prediction (NWP) forecast-27 ing fog and low visibility conditions is difficult since these require an accurate representation 28 of orography, surface, boundary-layer fluxes and inversions in the vertical temperature profile 29 (Stull, 2000; Jacobs et al., 2008). Operational forecasting of temperature inversions depends 30 on the availability of suitable observations (Roach et al., 1976; Jacobs et al., 2005; Fowler 31 et al., 2011) to locate the inversion. For example high-frequency reporting of vertical pro-32 files of temperature and wind may provide extra information for use in NWP assimilation and 33 nowcasting (Dance, 2004; Rennie et al., 2011; Simonin et al., 2014; Sun et al., 2014; Ballard 34 et al., 2015; James and Benjamin, 2017). Furthermore several authors (de Haan and Stoffelen, 35 2012; de Haan, 2013; Strajnar et al., 2015; Lange and Janjic, 2016) have demonstrated positive 36 impacts in regional NWP models when assimilating derived observations from aircraft reports 37 using Mode-Selective (Mode-S) Enhanced Surveillance (EHS), a system which transmits bi-38 nary coded messages to an aircraft's transponder and receives binary coded replies (Boisvert 39 and Orlando, 1993; ICAO, 2010). 40

Strajnar et al. (2015, figure 7) showed that meteorological routine air reports (MRAR) of ambi-41 ent temperature, obtained from the secondary surveillance radar (SSR) using Mode-S, centred 42 around Ljubljana airport, Slovenia, have a spatial and temporal resolution sufficient to locate a 43 temperature inversion at around 1000 m above the surface. However, direct reports of ambient 44 temperature using Mode-S MRAR is not routinely available since not all SSRs and not all air-45 craft are configured to make such reports. De Haan (2011) showed that Mode-S EHS reports 46 of Mach number and true-airspeed, centred around Schipol airport, Netherlands, could be used 47 to derive ambient temperature. In de Haan (2011, Figure 7) we noted that, after quality control 48 and smoothing, the derived ambient temperature from a single aircraft profile may also locate 49 temperature inversions. However, de Haan (2011); Mirza et al. (2016); Mirza (2017, table 6.2) 50

and Stone (2017) suggest that the uncertainty in the derived temperature from a single aircraft at low levels can range between 2 K and 10 K. This degree of uncertainty makes it difficult to locate the height and magnitude of the temperature inversion.

Stone and Kitchen (2015) showed that a mean temperature for a layer of thickness 2000 m could be computed using the global navigation satellite system's altitude reported by an aircraft's automatic dependent surveillance-broadcast (ADS-B) system. However, this method for determining thickness temperature is too coarse to resolve a temperature inversion.

58 All these methods use Mode-S/ADS-B reports from single aircraft to obtain temperature ob-

<sup>59</sup> servations. In our new work, we investigate the usefulness of using all available Mode-S EHS

<sup>60</sup> reports from multiple aircraft to estimate a vertical temperature profile.

In section 2 the current methods for obtaining in situ temperature measurements are described. 61 Section 3 describes the method used to collect Mode-S EHS reports, how the Mach temperature 62 observation is derived, and how these are aggregated to form a mean temperature observation. 63 Section 4 defines four smoothing filters used to reduce the variance in Mode-S EHS reports. 64 These are centred moving average, block average, linear regression and irregular exponential 65 smoothing. In section 5 we apply the method described in section 3 to a case study based 66 around London Heathrow to indicate the presence of temperature inversions. In section 6 we 67 apply the four low-pass filters, to a sample of the data for the London Heathrow domain. In 68 section 7 we show that the aggregated mean temperature profiles may provide useful informa-69 tion for operational meteorology, at least until temperature reports by ADS-B become more 70 routinely available (RTCA, 2012). All times are expressed as Universal Time Coordinated 71 (UTC). 72

# **73 2** In situ Upper Air Temperature Observations.

In situ observations of upper air temperature are made using a temperature sensor fixed to
a device which ascends or descends between the surface and the top of the troposphere or
beyond. Two types of such devices are the radiosonde and commercial aircraft.

For operational meteorology, modern radiosondes sample the atmosphere every second during ascent (World Meteorological Organisation, 2014, Ch 12, p.348), which can take up to two hours. Typically, radiosondes are launched from fixed sites that are widely separated (approximately 100 km) and report at fixed times (usually 0000 and 1200 UTC) so do not provide sufficient horizontal spatial or temporal resolution to capture the onset or duration of a temperature inversion (Fowler, 2010).

The common method of receiving observations from commercial aircraft is from the Aircraft 83 Meteorological Data Relay (AMDAR) program. An AMDAR equipped aircraft reports the 84 horizontal wind and ambient temperature obtained from the aircraft's flight management sys-85 tem (Painting, 2003). These reports are compiled on-board the aircraft and are transmitted to 86 a ground station. The frequency of transmission depends on the phase of flight (and whether 87 the aircraft is configured to send a report). For example, an aircraft may be configured to re-88 port every 6 seconds for the first 90 seconds during ascent then every 20 seconds until level 89 flight; during level-flight reports are every 3 to 10 minutes; during descent reports are every 90 60 seconds (Painting, 2003, p.32). 91 In Europe, the AMDAR program is managed by E-AMDAR which provides at least one ver-

In Europe, the AMDAR program is managed by E-AMDAR which provides at least one vertical profile once every three hours to participating National Meteorological Services (NMS)
from around 100 airports across Europe. The Met Office obtains one vertical profile once every hour at major airports. In Europe and the UK, the reporting frequency of vertical profiles
depends on the financial resources made available by the NMS. This contrasts with Air Traffic
Management (ATM) which can interrogate an aircraft's transponder at a much higher frequency
from a ground station SSR.

# **3** Aggregation of Mode-S EHS Reports.

Mode-S EHS is used by ATM to retrieve routine reports on an aircraft's state vector at a high temporal frequency (every 4 to 12 seconds). The aircraft's state vector consists of true-airspeed (hereafter referred to as the airpseed), magnetic-heading, ground speed, ground heading, altitude and Mach number. These Mode-S EHS reports can be used to derive estimates of the ambient air temperature and horizontal wind at the aircraft's location (de Haan, 2011).

During the study period, the Met Office used a Mode-S EHS receiver network which consists of five receivers (Stone and Pearce, 2016). Reports that are actively polled for by ATM and those routinely broadcast by aircraft are collected and processed by the Met Office receiver network.

The Met Office Mode-S EHS receivers are co-located at sites used for the weather radar network, which provide a good line of sight of aircraft flying above 500 m, power supply and communication network. The Mode-S EHS reports are collated then transmitted in batches every 10 minutes to a central processing facility, where the data are then passed through a quality control process (Stone and Pearce, 2016; Mirza, 2017). However, this network of Mode-S EHS receivers may be sub-optimal for the acquisition of Mode-S EHS reports at low levels,



Figure 1: Spatial distribution of trajectories (circles, colour-coded by altitude) for ascending and descending aircraft within the London Heathrow domain, derived from Mode-S EHS reports received between 1200 to 1300 on 4 January 2015. The domain extends for a distance of 80 km east-west, 40 km north-south, height 3000 m from the surface, with London Heathrow airport at the domain's centre. Points where the aircraft's roll angle is greater than 5°, i.e. when turning, are removed since these data are considered unreliable. (Cartography ©OpenStreetMap contributors, licensed as CC BY-SA https://www.openstreetmap.org/copyright, 2018)

e.g., below 500 m, due to loss of the line of sight required to receive Mode-S EHS reports.

Figures 1 and 8 (see supplementary section) show the distribution of the Mode-S EHS reports 116 received from the Met Office Mode-S EHS receivers for a domain centred around London 117 Heathrow airport. The domain's dimensions are sufficient to contain the trajectories of aircraft 118 arriving at or departing from London Heathrow. Trajectories for descending aircraft are longer 119 than for ascending aircraft. The domain excludes the areas where aircraft are held prior to 120 their descent. The domain is not cuboid but can be imagined as an inverted truncated pyramid, 121 centred at the airport. (In the supplementary section, figure 9 shows the distribution of Mode-S 122 EHS reports for a domain centred around London Gatwick airport.) 123

The Mach Temperature,  $T_{MACH}$ , is derived from Mode-S EHS reports of Mach number, Mand airspeed,  $V_A$ , (de Haan, 2011; Mirza *et al.*, 2016), such that

$$T_{\text{MACH}} = \frac{T_0}{A_0^2} \left[ \frac{V_A}{M} \right]^2, \tag{1}$$

where the speed-of-sound  $A_0 = 340.294 \text{ ms}^{-1}$  and the assumed surface temperature  $T_0 = 288.15 \text{ K}$ , are reference values defined at mean-sea-level pressure under international standard atmosphere conditions (ICAO, 1993).

To use as many of the Mode-S EHS reported data as possible they are aggregated to form a 129 mean Mach Temperature,  $\overline{T}_{MACH}$ , observation. This 'aggregated observation' (Mirza *et al.*, 130 2016; Mirza, 2017, Ch3) is the arithmetic mean of all the Mach Temperatures, derived using 131 equation (1), for all Mode-S EHS reports received within a defined time period and in a spec-132 ified horizontal layer. The assigned position of  $\overline{T}_{MACH}$  is set at the centre of the horizontal 133 layer and at the mean pressure altitude of all the reporting aircraft within. These layers form 134 a vertical profile of  $\overline{T}_{MACH}$  observations when stacked in the vertical, which is centred around 135 an airport. 136

We treat the errors as random so that the aggregated observation has a smaller error than an individual observation, since if the errors are random and uncorrelated then the standard error of the mean scales by  $1/\sqrt{n}$ , where *n* is the number of reports (Hoel, 1984, Ch 5 and Ch 10).

# **4 Temporal smoothing using low-pass filters**

Studies by de Haan (2011); Mirza et al. (2016) have shown that Mach number and airspeed in 141 equation (1) are subject to fluctuations which result in unrealistic values of derived tempera-142 ture. These fluctuations are thought to arise as a result of the reduced precision of these data 143 caused by the Mode-S EHS transponder processing the data prior to its transmission. De Haan 144 (2011) showed that by applying a linear smoothing algorithm to the time series of Mode-S 145 EHS reported Mach number and airspeed of a single aircraft before computing the derived 146 Mach Temperature then the large fluctuations in the latter are reduced. This action of linear 147 smoothing is similar to that of a low-pass filter, which reduces high-frequency components of 148 a time-varying signal. We apply and evaluate a selection of low-pass filters. 149

The low-pass filters described in this section are applied to the time series of Mode-S EHS reports for each aircraft trajectory and the result of the low-pass filter is used to generate a new aircraft report. Using this filtered time series of reports the Mach Temperature report is
 recomputed.

In our description of the filters, we use the notation  $x_k$ , for the value of an individual Mode-S

EHS report, with assigned time  $t_k$ . The filtered reports,  $X_{\bar{t}}$ , are computed by averaging over a validation window of length  $W_L$ , and they are assigned a validity time,  $\bar{t}$ .

#### 157 4.1 Block-window average (BLK)

The block-window average method creates a time series of Mode-S EHS reports using the average of all reports within a validation window, of length  $W_L$ . The time series is split into a sequence of non-overlapping blocks then the average of each block is computed. In computing the average no report is used more than once. The newly filtered time series is given by,

$$X_{\bar{t}} = \frac{1}{2m+1} \sum_{j=-m}^{+m} x_{k+j} \text{ for } k = m+1, 3m+2, 5m+3, \dots, \left\lfloor \frac{N}{2m+1} \right\rfloor (2m+1) - m, (2)$$

where N is the total number of reports in the time series and  $\lfloor \frac{N}{2m+1} \rfloor$  is the number of validation windows of length  $W_L = 2m + 1$  in the dataset. (The floor operator  $\lfloor z \rfloor$ , gives the greatest integer that is less than or equal to z (Oldham *et al.*, 2010, p.68).) The validity time,  $\bar{t}$ , is given by,

$$\bar{t} = \frac{1}{2m+1} \sum_{j=-m}^{+m} t_{k+j}.$$
(3)

This method is simple to implement but is not robust. It is susceptible to large variations since all the reports within the validation window are equally weighted.

#### 160 4.2 Centred moving average (CMA)

This is a straightforward method of computing a value over a short window length,  $W_L = 2m + 1$ . This method is also known by other names, e.g., running-mean, running-average, sliding-window average. Our method uses m reports before and after the current report, which is at the centre of the window. Each report is weighted equally, so reports from the start to the end of the window are treated to be of the same importance (Savitzky and Golay, 1964; Wendisch and Brenguier, 2013). The new time series is given by,

$$X_{\overline{t}} = \frac{1}{2m+1} \sum_{j=-m}^{+m} x_{k+j} \quad \text{for } k = m+1, m+2, m+3, \dots, N-m,$$
(4)

with the validity time given by eq. (3).

168 However, this method is also not robust since it can be affected by large outliers, and fluctua-

tions in the new time series may lag behind those seen in the original time series, although themagnitude of the variations is reduced.

#### 171 4.3 Piece-wise linear regression (LIN)

This uses the least squares regression method to compute a local rate of change, which is assumed to be linear over the validation window,  $W_L$ . In other words, the mean values obtained from fitting a straight line to the data locally are used to create the new time series. This is a statistical method that minimises the differences between a control variable and predicted values. The new time series is given by

$$X_{\overline{t}} = \alpha \overline{t} + \beta, \tag{5}$$

where the validity time is given by eq. (3). The local constant,  $\beta$ , is defined as

$$\beta = \overline{x} - \alpha \overline{t}.\tag{6}$$

where

$$\overline{x} = \frac{1}{2m+1} \sum_{j=-m}^{+m} x_{k+j}, \text{ for } k = m+1, m+2, m+3, \dots, N-m,$$
(7)

i.e., the local mean  $\overline{x}$  computed over the window. The corresponding local rate of change,  $\alpha$ , (i.e., the gradient) is given by,

$$\alpha = \frac{\sum_{j=-m}^{+m} (x_{k+j} - \overline{x})(t_{k+j} - \overline{t})}{\sum_{j=-m}^{+m} (t_{k+j} - \overline{t})^2}.$$
(8)

<sup>172</sup> Unlike the centred moving average this method is more responsive to variations in the time <sup>173</sup> series.

### 174 4.4 Irregular exponential moving average (IRR)

The exponential smoothing method is similar to the centred moving average except observations are weighted according to their position in time. The current observation is weighted more than the observations made at earlier times. The simple exponential moving average (Brown, 2004; Kim and Huh, 2011) assumes observations are available at regular time intervals. However, since the Mode-S EHS reports used to construct aircraft trajectories may be at irregular

time intervals and there may be missing data, the Wright (1986) method is used, which extends

the exponential smoothing method to irregular time intervals. The new time series is given by,

$$X_{t_k} = (1 - V_k) X_{t_k - 1} + V_k x_{t_k}, \tag{9}$$

where

$$V_k = \frac{V_{k-1}}{b_k + V_{k-1}}$$
(10)

and

$$b_k = (1-a)^{(t_k - t_{k-1})},\tag{11}$$

182 for k = 2, 3, 4, ..., N, and  $0 \le a < 1$ .

The value a is a smoothing parameter which determines the proportion of the new information to be added to the running average. The parameter  $V_k$  is a weighting function which is given an initial value of  $V_1 = 1$ . The larger the value of the parameter  $V_k$ , the less weight is given to the running average. The weighting function depends on the time separation between reports. For each  $X_{t_k}$  the assigned validity time is  $t_k$  since the former directly replaces each  $x_{t_k}$ .

#### **188 4.5 Consistency check**

We apply a consistency check so that the horizontal spatial and temporal resolutions of the time series are reasonably consistent along the aircraft trajectory. This consistency check is applied because there are fewer Mode-S EHS reports along an aircraft's trajectory than are actually available in principle.

We assume that a break in the time series of reports arise as a result of either (a) the aircraft exiting from a turning point on its approach to land, (b) that it passed out of then re-entered the airport domain, shown in figure 1, (c) that the aircraft was not within the line of sight reception to the Mode-S EHS receiver or (d) due to quality control pre-processing of Mode-S EHS reports, performed at the monitoring site (Stone and Pearce, 2016), which removes reports when an aircraft's roll angle exceeds 5 degrees creating gaps in the time series of reports.

<sup>199</sup> The consistency check is used to determine when a low-pass filter outputs a filtered value. <sup>200</sup> The filtered value  $X_{\bar{t}}$  is set to a missing data indicator when the time difference between two <sup>201</sup> successive reports,  $\delta t$ , used to compute the filtered value is greater than a maximum permitted <sup>202</sup> time difference,  $\delta t > \delta t_{max}$ . (This affects the BLK low-pass filter more as reports are only <sup>203</sup> used once.) The value of  $\delta t_{max}$  ensures that the data input to the low-pass filter are closely related in time and space.

We select a value for  $\delta t_{max}$  equal to the standard deviation of the time difference between successive Mode-S EHS reports along an aircraft's trajectory. For the selected day we use all aircraft trajectories to compute this standard deviation. The result is rounded to the nearest whole second.

The effect of applying the consistency check is to set the maximum time window for sampling the meteorological conditions based on the validation window of length  $W_L$ .

# 211 5 Inversion Case Study

In this section, we use a case study to identify useful meteorological information for the London Heathrow domain between 4 and 5 January 2015. This period was chosen because fog was a persistent weather feature. One of the meteorological conditions for fog to arise is the presence of a temperature inversion at low altitude or near the surface.

#### 216 5.1 Observations

To assess the information content of the  $\overline{T}_{MACH}$  vertical profile we compare it to temperature 217 reports from other observation systems. We use the high-resolution temperature profile from 218 Herstmonceux, the nearest radiosonde station. We also use AMDAR temperature reports. We 219 note also that all AMDAR reporting aircraft also report Mode-S EHS. We assume that ra-220 diosonde and AMDAR observations are representative of the meteorological conditions. The 221 vertical profile of  $\overline{T}_{MACH}$  is compared to the forecast mean vertical temperature profile from the 222 Met Office's limited-area, high-resolution, convection-permitting NWP model for the United 223 Kingdom, the UKV (Lean et al., 2008; Tang et al., 2013); the mean is calculated using UKV 224 vertical profiles at selected points across the London Heathrow domain. We note that the ra-225 diosonde and AMDAR temperature reports that we use for comparison are not assimilated by 226 the UKV. 227

In figure 2 we show all temperature reports for the London Heathrow domain on 4 January 2015 with a validity time of 0600, that is all observations received between 0530 and 0630. The  $\overline{T}_{MACH}$  profile (black triangles) is constructed using the aggregation method described in section 3. The  $\overline{T}_{MACH}$  error bars (black) are the 95% confidence limits for the mean using the Student-t distribution (Hoel, 1984, Ch 5 and Ch 11). For comparison, we show in situ observations from two other observing systems: radiosonde and AMDAR. The radiosonde was launched at 0515, headed due south of its launch site at Herstmonceux and reached an altitude



Figure 2: Temperature reports for the London Heathrow domain on 4 January 2015 for the period 0530 to 0630. Aggregated Mach Temperature observation and its 95% confidence interval (black triangles), with the number of Mode-S EHS reports used shown by the horizontal bars from the left, and the number of reporting aircraft shown on the right-axis. Herstmonceux radiosonde report valid at 0600 (black solid line, with its reported precision of  $\pm$  0.5 K shown by the grey shading), AMDAR reports (large diamonds) and their reported precision of  $\pm$  0.5 K (error bars), and the mean UKV forecast and its 95% confidence interval, valid at 0600 from the forecast run at 0300 4 January 2015 (narrow diamonds).

- of 3000 m at 0524. Position and temperature reports were made every 2 s. The region of the
- atmosphere sampled by the radiosonde is not contained within the London Heathrow domain.
   AMDAR temperature reports are shown as point observations (Painting, 2003), received be-
- tween 0557 and 0617 from an aircraft destined to land at London Heathrow during this period.
- <sup>239</sup> We also show the mean UKV forecast temperature profile for the London Heathrow domain

with the validity time 0600. The mean forecast temperature profile is computed by using a

sample of nine 1-D column profiles from across the London Heathrow domain (Mirza, 2017,

Fig 5.5). The standard deviation of the mean forecast temperature profile indicates that at this

time between the pressure altitude range 300 m and 3000 m there is little variation across the

domain (<0.5 K) and below 300 m it is around 1.5 K. (For this pressure altitude range In-

gleby and Edwards (2014) estimated that the average UKV model error to be  $\pm 0.75$  K when

<sup>246</sup> compared against high-resolution radiosonde reports.)

#### 247 5.2 Observed Meteorological Features

In figure 2 the radiosonde report indicates the presence of two temperature inversions: a lowlevel temperature inversion between 500 m and 900 m, reported at 0516, and an elevated temperature inversion between 1800 m and 2000 m, reported at 0520. The AMDAR observations, reported between 0557 and 0612, are broadly in agreement with the radiosonde. These in situ observations provide a broad description of the vertical temperature structure of the atmosphere between Heathrow airport and Herstmonceux. However, there is a clear difference between these in situ observations and the mean UKV forecast for the London Heathrow domain.

The UKV at 0600 forecasts a low-level inversion between the surface and 300 m but does not forecast the elevated inversion between 1800 m and 2000 m. However, the  $\overline{T}_{MACH}$  observations, obtained between 0530 and 0630, do suggest that an elevated inversion is present.

The radiosonde and AMDAR reports were not included in the UKV analysis (i.e., the initial 258 state of the NWP model) as they were received after the data assimilation observation pro-259 cessing period, 0130 to 0419. Therefore the UKV forecast will not have taken into account 260 the existence and the location of the temperature inversions shown by these observations and 261 there are no other sources of in situ upper air temperature observations during the observation 262 processing period. Furthermore, the elevated temperature inversion is not forecast by the UKV 263 at 0300, 0400 and 0500 within the London Heathrow domain, but this may also be due to 264 deficiencies in the physical modelling within the UKV. 265

The  $\overline{T}_{MACH}$  observations appear consistent with the radiosonde and AMDAR reports between 700 m and 3000 m. In this case, while there are insufficient AMDAR reports to resolve the inversion, its presence is shown by the  $\overline{T}_{MACH}$  observations at around 1900 m, even though the magnitude of the inversion suggested by the  $\overline{T}_{MACH}$  report differs significantly from that shown by the radiosonde. The radiosonde and AMDAR show the inversion to be higher, but this difference could be accounted for by a horizontal variation in the inversion height. Below 700 m the  $\overline{T}_{MACH}$  observations are more consistent with the UKV forecast, except around <sup>273</sup> 300 m, where the difference between the UKV and  $\overline{T}_{MACH}$  is of the same magnitude as at <sup>274</sup> 2000 m, i.e., approximately 5 K.

The absence of the elevated temperature inversion at around 2000 m in the UKV forecast would 275 be important for the subsequent forecasts of other meteorological phenomena. An elevated in-276 version in effect caps vertical movement and dispersion of atmospheric aerosols. This may 277 affect the forecast conditions for solar insolation and the formation or persistence of fog and 278 cloud (Fowler *et al.*, 2011). We suggest that  $\overline{T}_{MACH}$  observations could provide an additional 279 source of information, albeit a qualitative source, on the vertical temperature profile that may 280 otherwise be unknown, since the 0600 Herstmonceux radiosonde report is made only on de-281 mand (unlike the reports at 0000 and 1200). We illustrate the qualitative information contained 282 in the  $\overline{T}_{MACH}$  observations in figure 3. 283

Figure 3 shows the temperature reports available for the validity time 0900 on 4 January 2015; 284 these are all reports received between 0830 and 0930. There are no in situ observations from 285 radiosonde because there is no routine launch at this time of day. The 13 AMDAR observations 286 were reported between 0830 to 0837 from an aircraft on a descent path to Heathrow airport. The 287 computation and depiction of the  $\overline{T}_{MACH}$  observations and UKV vertical temperature profile are 288 as described in figure 2. We note that  $\overline{T}_{MACH}$  observations suggest that the elevated inversion 289 noted in figure 2 still persists although at a lower altitude, between 1500 m and 1800 m, with 290 a broadly isothermal region between 1000 m and 1500 m. The AMDAR reports are broadly in 291 agreement with the presence of the temperature inversion but not with the isothermal region. 292 The UKV forecast for these two regions does not show either meteorological feature. The 293 AMDAR reports would not have been available for assimilation into the UKV. Figure 4 shows 294 the same time period but 24 hours later for which there are no AMDAR or radiosonde reports. 295 In this case, the UKV forecast and the  $\overline{T}_{MACH}$  observations show some agreement indicating 296 the presence of an elevated temperature inversion between 1000 m and 1500 m. Thus, in the 297 absence of other in-situ observations, the  $\overline{T}_{MACH}$  observations could provide useful information 298 about the vertical structure of the atmospheric temperature. 299

Figures 2, 3 and 4 all show that  $\overline{T}_{MACH}$  indicates warmer conditions compared to the UKV forecast. This may be due to a bias in  $\overline{T}_{MACH}$  resulting from the numbers of aircraft that are ascending and descending at any given time (although it is also possible that the UKV NWP model is biased). Studies by Mirza (2017) and Stone (2017) suggest that  $\overline{T}_{MACH}$  reports between the surface and 3000 m appear cooler than the ambient conditions when aircraft ascend, while for descents these reports appear warmer. These effects may be the result of aircraft manoeuvrings during ascent or descent. For example, most descending aircraft extend their



Figure 3: Temperature reports for the London Heathrow domain on 4 January 2015 for the period 0830 to 0930. Symbols are as described in figure 2. This plot shows the aggregated Mach Temperature reports and the corresponding number of Mode-S EHS reports, AMDAR reports, and the mean UKV forecast valid at 0900.

landing gear and set full flaps at a height of around 300 m. This causes a strong deceleration, 307 which could explain major deviations of the reported Mach number from the observed airspeed 308 and thus erroneous temperatures. In addition, the height where the  $\overline{T}_{MACH}$  profile deviates from 309 the other data coincides with the bottom (and the most probably populated) level of London 310 Heathrow's holding patterns at 2000 m. Aircraft on hold do significantly more manoeuvring 311 which may lead to a decrease in the accuracy of the derived  $T_{MACH}$  reports. Mirza et al. (2016, 312 Figure 11) suggest that with sufficient Mode-S EHS reports from a single aircraft type, e.g., 313 greater than 100 at each altitude interval, then any bias may be reduced to near zero. However, 314 Stone (2017, Figure 1b) suggests that the bias may depend on whether the aircraft is ascending 315

![](_page_16_Figure_0.jpeg)

Figure 4: Temperature reports for the London Heathrow domain on 5 January 2015 for the period 0830 to 0930. Symbols are as described in figure 2. This plot shows the aggregated Mach Temperature reports and the corresponding number of Mode-S EHS reports and the mean UKV forecast valid at 0900. The lowest two points (not shown) are  $283.4\pm3.6$  K and  $285.3\pm4.9$  K. There were no radiosonde or AMDAR reports available for this time period and altitude range.

- <sup>316</sup> or descending. Further research is needed to understand these effects, for example. a much <sup>317</sup> longer study such as was done for AMDAR (Drue *et al.*, 2008).
- <sup>318</sup> Figure 5 shows similar temperature reports as shown in figure 4 but for the validity time at
- 2100 on 5 January 2015; these are all reports received between 2030 and 2130. There are no
- radiosonde observations, but there were 9 AMDAR reports received between 2043 and 2045
- <sup>321</sup> from an aircraft departing from Heathrow. The UKV mean profile is for 2100 from the forecast
- run at 2100 on 5 January 2015, so this represents the NWP analysis. Unlike the previous

![](_page_17_Figure_0.jpeg)

Figure 5: Temperature reports for the London Heathrow domain on 5 January 2015 for the period 2030 to 2130. Symbols are as described in figure 2. This plot shows the aggregated Mach Temperature reports and the corresponding number of Mode-S EHS reports, available AMDAR reports and the mean UKV forecast valid at 2100 from forecast run at 2100 on 5 January 2015. The lowest two points (not shown) are  $283.0\pm1.9$  K and  $285.8\pm4.3$  K. There were no radiosonde reports available for this time period and altitude range.

examples, it is likely that the AMDAR reports were received in time for their assimilation prior to the UKV forecast run. Therefore there is a good correspondence between the AMDAR temperature reports and UKV mean temperature profile. The  $\overline{T}_{MACH}$  observations between 600 m and 3000 m also show a good correspondence, in particular capturing the elevated inversion between 900 m and 1500 m. However, in each of the cases shown, at or below 1000 m the  $\overline{T}_{MACH}$  observations show increased level of uncertainty, as shown by the 95% confidence limits, and large differences between the AMDAR and radiosonde observations, and the UKV forecasts.

The radiosonde and AMDAR reports are effectively instantaneous values, reporting on a time 331 scale of seconds to minutes. The  $\overline{T}_{MACH}$  observation uses all available Mode-S EHS reports 332 over a large spatial domain, and is an average over the hour thus representing the mean condi-333 tions in space and time. The horizontal bars shown in figure 2 indicate the number of observa-334 tions used to compute each  $\overline{T}_{MACH}$  observation. The mean time difference between reports is 335 2 seconds per aircraft which corresponds to a horizontal spatial sampling scale around 250 m, 336 however, any variability on this scale will be lost due to the averaging process. Where there is 337 an agreement between the  $\overline{T}_{MACH}$  observation and the UKV this may be due to the latter also 338 representing the mean conditions over the hour, although its spatial sampling scale is 1500 m. 339

We do note that  $\overline{T}_{MACH}$  observations show a degree of variability, as represented by the 95% 340 confidence limits. The large variation in the computed  $\overline{T}_{MACH}$  observations may be due to the 341 low precision of the underlying data, mainly the Mach number (de Haan, 2011; Mirza, 2017). 342 The large uncertainty in the confidence limits is due to the drop in the available number of 343 Mode-S EHS reports used to compute the  $\overline{T}_{MACH}$  observations. Using the Student-t distribu-344 tion to compute the confidence limits may be unreliable or unsuitable at these low levels as 345 the distribution of the individual  $T_{MACH}$  reports becomes multi-modal. Since the atmospheric 346 conditions do not appear to vary greatly over the hour, we suggest that variability of  $\overline{T}_{MACH}$  ob-347 servations is likely to be due to the precision of the Mode-S EHS data used to derive the Mach 348 Temperature (Mirza et al., 2016; Mirza, 2017). This results in the poor characterisation of the 349 vertical temperature profile at levels below 1000 m. (Figures 10 and 11 in the supplementary 350 section (available online) show examples of the derived profiles for a similar size domain with 351 London Gatwick airport at its centre for the same case study period.) 352

This variability is not seen in the radiosonde and AMDAR reports, especially at levels below 1000 m. However, there are insufficient AMDAR reports to characterise fully the vertical temperature profile, and so they may not capture inversions between the surface and 600 m. The low reporting of AMDAR may be due to operational constraints, e.g., availability of suitably equipped aircraft or cost constraints which limit reporting to a single aircraft.

# **358** 6 Temporal smoothing using low-pass filters

#### 359 6.1 Motivation for low-pass filtering

In section 5.2 it was shown that  $T_{MACH}$  reports are subject to a high degree of variability especially at altitudes below 1000 m. De Haan (2011) and Mirza (2017) suggest the variability is due to the effects of Mode-S EHS processing. In this section, we apply four methods that perform the function of a low-pass filter, described in section 4, to a sample of the data for the London Heathrow domain. The filters are applied to the time series of Mode-S EHS reports for each aircraft within the London Heathrow domain. The filters create a new time series of smoothed Mode-S EHS reports which are then used to compute  $T_{MACH}$  observations (eq. (1)).

Figure 6 has four panels. Each panel shows the same short time series of non-smoothed Mode-S 367 EHS reports (grey dots) for Mach number and airspeed, and over the period of one minute there 368 are 28 reports of each. The corresponding derived Mach Temperature ranges between 269 K 369 and 291 K. However, such a change in the ambient temperature in one minute is unrealistic. 370 De Haan (2011) suggests that this magnitude of change in Mach Temperature is due to the low 371 precision of the reported Mach number. Mirza (2017) shows that this is indeed the case but 372 then goes on to suggest that the variation in Mach Temperature is also due to the asynchronous 373 changes in the Mode-S EHS reports of Mach number and airspeed. Close examination of figure 374 6(a) shows the effects of low precision and asynchronous changes. 375

In figure 6(a)(i) the first six Mach number reports show there are two step changes of -0.004 376 while the airspeed remains constant, indicated by region A, figure 6(a)(ii). These step changes 377 represent the reporting precision of the Mach number after Mode-S EHS processing. The 378 corresponding Mach Temperature, figure 6(a)(iii), computed using equation (1), show step 379 changes of +7 K. These changes occurred over 9 s with two step changes in altitude: 1821 m 380 to 1814 m to 1806 m (not shown). Equation (1) suggests that if the airspeed is constant then 381 a decrease in the Mach number corresponds to an increase in Mach Temperature. This is also 382 suggested by figure 2 where for the altitude range of these Mode-S EHS reports the radiosonde 383 and AMDAR reports indicate the presence of a temperature inversion. 384

In figure 6(a)(ii), the report at region B for airspeed shows a large step change of -8 knots while the Mach number and altitude are unchanged. This results in a step change of -21 K in the corresponding Mach Temperature in 1 s. Equation (1) suggests that if the Mach number is constant then a decrease in airspeed corresponds to a decrease in the Mach Temperature. However, for the 1 s over which this change takes place the aircraft's reported altitude remained at 1806 m and its horizontal displacement was 138 m. It is unlikely that the actual ambient temperature would change by this magnitude over such a short distance and time. But if we assume that temperature is constant then equation (1) shows that a decrease in airspeed should show a corresponding decrease in Mach number, which in this instance did not occur. We suggest, therefore, that the Mode-S EHS processing causes asynchronous changes in the Mach

<sup>395</sup> number and airspeed which may result in the observed large fluctuations in Mach Temperature.

Regions C and D show a synchronous change in Mach number and airspeed, which results in a change of Mach Temperature of -9.5 K. The changes in altitude for each occurrence were 1783 m to 1768 m over 5 s and 1737 m to 1722 m over 4 s. We suggest that the change in magnitude, while smaller than for the asynchronous case at region B, is due to the Mode-S EHS processing which reduces the precision of the Mach number and airspeed.

In summary, there are two effects of Mode-S EHS processing that may account for the observed variability in the derived Mach Temperature: the reduced precision of the reported Mach number and airspeed and their asynchronous changes. The use of a suitable low-pass filter may smooth out the step changes in Mach number and airspeed thus reducing the observed variability in the derived Mach Temperature. We consider the use of low-pass filters in the next section.

### 407 6.2 Applying low-pass filters to time series of Mode-S EHS Reports

We now explain how we set-up and use the low-pass filters. For the London Heathrow domain, 408 the consistency check  $\delta t_{max}$  is 6 s. For BLK (eq. (2)), CMA (eq. (4)) and LIN filters (eq. 409 (5)) the validation window is set with m = 2. This provides five reports for the validation 410 window, i.e., where each filtered report has two reports either side, which are used to compute 411 the mean value, except at the start and end of the time series. If 6 s is the maximum time 412 separation between the five reports within the validity window then the filtered report represents 413 the meteorological conditions sampled over 30 s. This is an appropriate sample time given 414 that aircraft are changing position horizontally and vertically. Typical ascent rates are 5-10 415  $ms^{-1}$  so a 30 s averaging could be over 150-300 m in the vertical. This is similar to the 416 vertical grid length in many NWP models. Typical glide speed would be 100-120 ms<sup>-1</sup> giving 417 a horizontal representation over 3.0-3.6 km. During the sampling time the aircraft may make 418 control movements that increases or decreases its altitude during any part of its phase of flight: 419 ascent, en-route or descent. These may be considered as an additional source of high-frequency 420 noise. 421

There is a trade-off between the parameters  $\delta t_{max}$  and m. If  $\delta t_{max}$  is too short in time then high-frequency components may not be sufficiently damped. Furthermore, this limits the num-

- <sup>424</sup> ber of reports used due to failing the consistency check (see section 4.5). If the window length is
- too large then over-smoothing may result which may cause the position and altitude of the tem-
- <sup>426</sup> perature inversion to be either misplaced or not detected. However, these parameters could be
- tuned for particular operational conditions at airports or to apply different consistency checks
- <sup>428</sup> for ascending and descending aircraft since rates of ascent are larger than rates of descent. The
- additional outputs of these low-pass filters (except IRR) are the means of the time, latitude,

430 longitude and pressure altitude quantities within the validation window.

- For IRR (eq. (9)), we use a smoothing factor a = 0.2. The weighting function (eq. (11)) is initialised with the time difference  $t_k - t_{k-1} = 1 s$ . These parameters were selected so that when the time separation between reports is 4 s, the expected SSR rotation rate, then the exponential smoothing will weight the previous filter value and the current observation equally. Thus the IRR low-pass filter replaces each Mode-S EHS report in the aircraft's trajectory, therefore, the
- <sup>436</sup> low-pass-filtered trajectory contains the same number of reports.

### 437 6.3 Effect of applying low-pass filters

In figure 6 the resulting smoothed Mach number, airspeed and recomputed Mach temperature are shown as the square points after applying the low-pass-filters discussed in section 4. The main effect of the low-pass filters IRR, CMA, and LIN (figures 6(b), 6(c), 6(d) respectively) is to smooth the step transitions in Mach number and airspeed which reduces the variance of the Mach Temperature distributions at each altitude bin. This is the desired effect as it shows that the impact of the high-frequency components is being diminished.

We apply each of these low-pass filter methods to all aircraft trajectories within the London 444 Heathrow domain. We then apply the aggregation method to recompute  $\overline{T}_{\rm MACH}$  for each hor-445 izontal layer (shown in figure 2). Figure 7(a)(i) shows the results after applying the different 446 low-pass filters. Figure 7(a)(ii) shows the difference between the smoothed and unsmoothed 447  $\overline{T}_{MACH}$  observations. Above 1000 m the difference ranges between  $\pm$  0.5 K. However, below 448 1000 m the magnitude of the smoothed  $\overline{T}_{MACH}$  is greater. The magnitude of the latter results 449 may arise because reports have been filtered out during the low-pass filtering. This is shown in 450 figure 7(b)(ii) where the number of reports for CMA and LIN are less than for IRR (the number 451 of reports for the unsmoothed profile is the same as for the IRR). The number of reports for 452 BLK low-pass filter is greatly reduced but this is expected since this method replaces a series of 453 reports with a single report whereas the other low-pass methods use substitution. The overall 454 effect of the applying the low-pass filters to the computed  $\overline{T}_{MACH}$  is minimal. However, the 455 low-pass filters have a greater effect on the computed standard deviation of the  $\overline{T}_{MACH}$ . 456

![](_page_22_Figure_0.jpeg)

Figure 6: Before (circles) and after effects (squares) of applying smoothing filters for one aircraft's time series of (i) Mach number and (ii) true-airspeed for (a) Block Average, (b) Irregular Exponential, (c) Centred Moving Average and (d) Linear Regression. (iii) Mach Temperature computed before and after smoothing.

![](_page_23_Figure_0.jpeg)

Figure 7: Effect of applying the different smoothing filters to Mode-S EHS reports of Mach number and true-airspeed along all aircraft tracks, London Heathrow domain, 4 January 2015 0530 to 0630. In each case (a) the resulting  $\overline{T}_{MACH}$  reports and (b) the estimated sample standard deviation are recomputed. Key:  $\blacksquare$  uncorrected  $\overline{T}_{MACH}$ , low-pass filtered:  $\blacktriangle$  BLK,  $\bullet$  CMA,  $\blacktriangledown$ , LIN,  $\blacklozenge$  IRR. Estimated error: $\blacksquare$  full precision,  $\bullet$  2 × quantisation,  $\blacktriangle$  quantisation.

Figure 7(b)(i) shows the effect of each low-pass filter on the computed standard deviation of the  $\overline{T}_{MACH}$ . For comparison also shown are the expected standard deviations for the  $\overline{T}_{MACH}$ , using the Mach Temperature error equation formulated by Mirza *et al.* (2016, Equation 16), assuming the following for the Mach number and airspeed: full precision error, precision due to quantisation error (Mirza *et al.*, 2016, figures 4 and 11) and precision due to double the quantisation error.

We used four low-pass filters: centred moving average (CMA), block average (BA), linear 463 regression (LR) and irregular exponential smoothing (IRR). For smoothing the time series of 464 reports above an altitude of 1000 m, the performance of each of the low-pass-filters was similar. 465 Below 1000 m there was a small difference between using the moving window methods and 466 the IRR. The former methods reduce variance more than the IRR. However, the advantage of 467 the IRR method is that it uses all the available reports whereas the moving window methods 468 removed reports as a result of the imposed quality control criterion. Furthermore, the IRR's 469 weighting function is time-dependent, giving most weight to the most recent datum. This may 470 reduce over-damping of high-frequency signals in the presence of a temperature inversion that 471 would otherwise be smoothed by the moving window methods. However, each of the methods 472 used to minimise the fluctuations in the Mode-S EHS derived observations, i.e., aggregation 473 and low-pass filtering, effectively reduce the space and time resolution of the data. 474

# 475 **7** Summary and Conclusions

This paper used Mode-S EHS reports exchanged between an aircraft and air traffic control to 476 derive Mach Temperature. Using an aggregation of Mach Temperature reports from all air-477 craft within a defined region of an airport, e.g., the London Heathrow domain, vertical profiles 478 of the mean Mach Temperatures,  $\overline{T}_{MACH}$ , for horizontal layers were constructed and used to 479 identify a meteorological feature, temperature inversion, which is important for operational 480 aviation weather forecasting and numerical weather prediction. To improve the representation 481 of  $\overline{T}_{MACH}$ , low-pass filters were applied to the time series of Mode-S EHS reports of Mach 482 number and airspeed for all aircraft within the London Heathrow domain. The low-pass fil-483 ter smoothed the discrete transitions of the Mach number and airspeed, which occur due to 484 their low precision. Anomalous values of the derived Mach Temperature, which arise due to 485 the asynchronous change between the Mach number and airspeed, were also smoothed. The 486 overall effect of the low-pass filter reduced the variance of the  $\overline{T}_{MACH}$  by as much as 50%. 487

We compared hourly  $\overline{T}_{MACH}$  profiles with in situ observations of temperature reported by radiosonde and AMDAR, when available. We found that the  $\overline{T}_{MACH}$  profile between 1000 m and

3000 m shows some agreement with these in situ observations whereas below 1000 m there 490 was less agreement, where the magnitude of the difference between the in situ observations 491 and the  $\overline{T}_{MACH}$  was as great as 6 K. In our comparisons (figures 2, 3,4 and S3),  $\overline{T}_{MACH}$  seems 492 to be in reasonable agreement with AMDAR and radiosonde data down to 600-700 m, a little 493 lower than the 1000 m limit that we conservatively estimated. However, the results also show 494 that some significant deviations can occur between 600 m and 1000 m. These arise in the early 495 morning and the late evening, when there are few aircraft and so fewer Mode-S EHS reports 496 at the lower levels. This scarcity may be due to the interruption of the line of sight between 497 the aircraft and the Mode-S EHS receiver station. Hence we chose 1000 m as a safe lower 498 limit for practical application. Daily operations may achieve better but this is best left to the 499 meteorologist's judgement as they gain experience with the application. 500

However, the comparison against in situ observations is difficult since these are point based values, measured on time-scales of seconds to minutes, compared with the hourly mean of the aggregated Mach Temperature. Moreover, the radiosonde observations are not located within the airport domains. The temperature differences observed below 1000 m are unlikely to be due to changes in the ambient temperature; nor the prevailing meteorological conditions at the surface on the day (near freezing conditions, low wind speed and fog) but more likely due to Mode-S EHS processing (de Haan, 2011; Mirza *et al.*, 2016; Mirza, 2017; Stone, 2017).

We also compared the hourly aggregated Mach Temperature against the UKV model forecasts. We found similar results to our comparison with in situ observations. Furthermore, we found that the Mach Temperature profiles identified regions where temperature inversions may be present but which were not present in the UKV forecast, thus showing that Mach Temperature profiles may provide additional information for use in NWP.

From analysing the time series of the Mode-S EHS reports, we found that the Mode-S EHS processing also results in step changes in the reports of Mach number and airspeed that are asynchronous in time. This results in very large fluctuations in the corresponding Mach Temperature, ranging from 5 K to 9 K between adjacent reports.

<sup>517</sup> We conclude that applying a low-pass filter to the time series reports of Mach number and air-<sup>518</sup> speed could be beneficial as a pre-processing step prior to NWP data assimilation but further <sup>519</sup> research would be needed in order to tune the filter parameters. Moreover, the IRR method <sup>520</sup> could be used as the basis for a Kalman filter. While the quantitative value of the mean Mach <sup>521</sup> Temperature may have a large uncertainty, the qualitative value of the constructed vertical pro-<sup>522</sup> file of the mean Mach Temperature may provide additional information that may be useful for <sup>523</sup> operational meteorology, e.g., identifying the possible locations for the occurrence of temperature inversions, when combined with other available sources of information. Furthermore, this may help aviation meteorologists to improve their forecasts for ATM by verifying in near-realtime the performance of the NWP forecast. However, further studies should be undertaken to assess this aspect.

The most common Mode-S EHS report is the aircraft's state vector from which temperature 528 and horizontal wind observations can be derived. However, an alternative to Mode-S EHS is 529 Mode-S MRAR (Strajnar, 2012; Strajnar et al., 2015), but the current regulatory environment 530 does not require aircraft or ATM to make such reports available. The technology and capability 531 already exist for the direct reporting by aircraft of the temperature and horizontal wind. There-532 fore, in the interest of making more effective use of aircraft based observations for operational 533 meteorology and numerical weather prediction, the aviation industry should be encouraged to 534 implement either Mode-S MRAR reporting or its planned successor ADS-B. 535

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## 544 Supplementary Section.

In this supplementary section, in figures 8 and 9 we show for comparison the spatial distribution 545 of Mode-S EHS reports for London Heathrow and London Gatwick, received between 1200 to 546 1300 on 4 January 2015. London Gatwick is located 40 km south east of London Heathrow 547 airport. At this time the air traffic flow was east to west, with aircraft arriving from the east 548 and departing to the west. Each domain extends for a distance of 80 km east-west, 40 km 549 north-south, height 3000 m from the surface, with the airport at the domain's centre. Points 550 where the aircraft's roll angle is greater than  $5^{\circ}$ , i.e. when turning, are removed since these 551 data are considered unreliable. While the domains appear to be cuboid this is not the case. The 552 sampled volume of space resembles an inverted truncated pyramid. Figures 10 and 11 we show 553 the vertical temperature profile for the London Gatwick domain for two separate time periods. 554 The method used to compute the  $\overline{T}_{MACH}$  observations is described in section 5.1. 555

In figure 10, as noted in section 5.1, the Herstmonceux (45 km south east of Gatwick) ra-556 diosonde temperature profile (black line) shows that temperature inversions are present. The 557 UKV temperature profile forecasts a low-level temperature inversion between 150 m and 300 m. 558 The  $\overline{T}_{MACH}$  observations suggest that the upper-level inversion is at 1600 m rather than around 559 2000 m shown by the radiosonde. Furthermore, the  $\overline{T}_{MACH}$  observations suggest that there 560 is an isothermal region between 800 m and 1600 m, which is not shown by the UKV fore-561 cast or radiosonde. We note that there were no AMDAR reports for this period and location. 562 The  $\overline{T}_{MACH}$  observations suggest that the rate of decay of the temperature inversion was much 563 slower than that shown by the UKV forecast. 564

In figure 11, as noted in section 5.1, the Herstmonceux radiosonde temperature profile (black line) shows that temperature inversions are present. The UKV forecasts similar temperature inversions, although lower down when compared with the radiosonde. For this period and location there were five AMDAR reports, however, these do not show clearly the location of the temperature inversions. The  $\overline{T}_{MACH}$  observations show clearly the presence of the upperlevel inversion but suggest it is lower down than forecast.

In both these cases, the  $\overline{T}_{MACH}$  observations at low levels may not be reliable because of the low number of Mode-S EHS reports used to make these report, as indicated by the width of the 95% confidence intervals, and the general increase in error at levels below 1000 m. Nonetheless, the  $\overline{T}_{MACH}$  observations may provide useful information when compared alongside other in situ temperature observations.

![](_page_28_Figure_0.jpeg)

Figure 8: Spatial distribution of trajectories (circles, colour-coded by altitude) for ascending and descending aircraft within the London Heathrow domain, derived from Mode-S EHS reports received between 1200 to 1300 on 4 January 2015. (Cartography ©OpenStreetMap contributors, licensed as CC BY-SA https://www.openstreetmap.org/copyright, 2018)

![](_page_29_Figure_0.jpeg)

Figure 9: Spatial distribution of trajectories (circles, colour-coded by altitude) for ascending and descending aircraft within the London Gatwick domain, derived from Mode-S EHS reports received between 1200 to 1300 on 4 January 2015. (Cartography ©OpenStreetMap contributors, licensed as CC BY-SA https://www.openstreetmap.org/copyright, 2018)

![](_page_30_Figure_0.jpeg)

Figure 10: London Gatwick 2015-01-04, Mode-S EHS aggregated Mach Temperature vertical profiles (triangles), radiosonde (black) and mean UKV (narrow diamonds) temperature profiles. Symbols are as described in figure 2.

![](_page_31_Figure_0.jpeg)

Figure 11: London Gatwick 2015-01-05, Mode-S EHS aggregated Mach Temperature vertical profiles (black triangles), radiosonde (black), available AMDAR reports (grey triangles) and mean UKV (narrow diamonds) temperature profiles. Symbols are as described in figure 2.

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